



- **Combining Game Strategies and Evolutionary Algorithms for CAD Parameterisation and Multi-Point Optimisation of Complex Aeronautic Systems**

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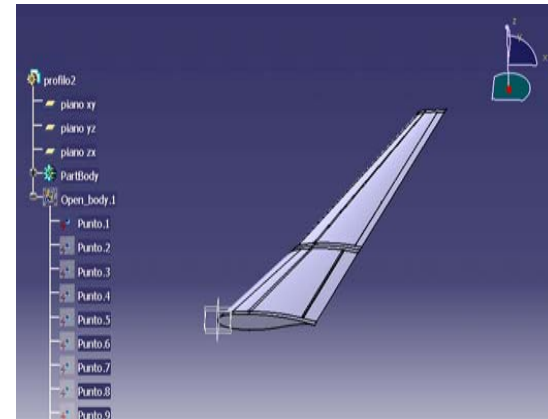
Jacques Periaux Dassault Aviation - Direction de la Prospective, St Cloud, Paris

Definition of the problem: multi-point 3D wing design

- Transonic point:

minimise **Drag**

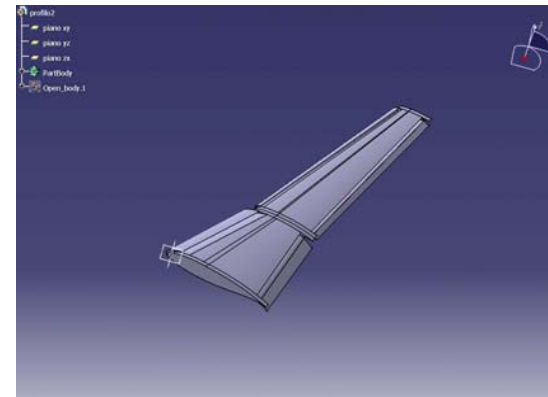
- angle of attack: 2°
- Mach number: 0.7
- lift and max thickness constrained to original



- Subsonic point:

maximise **Efficiency**

- angle of attack: 17.12°
- Mach number: 0.12

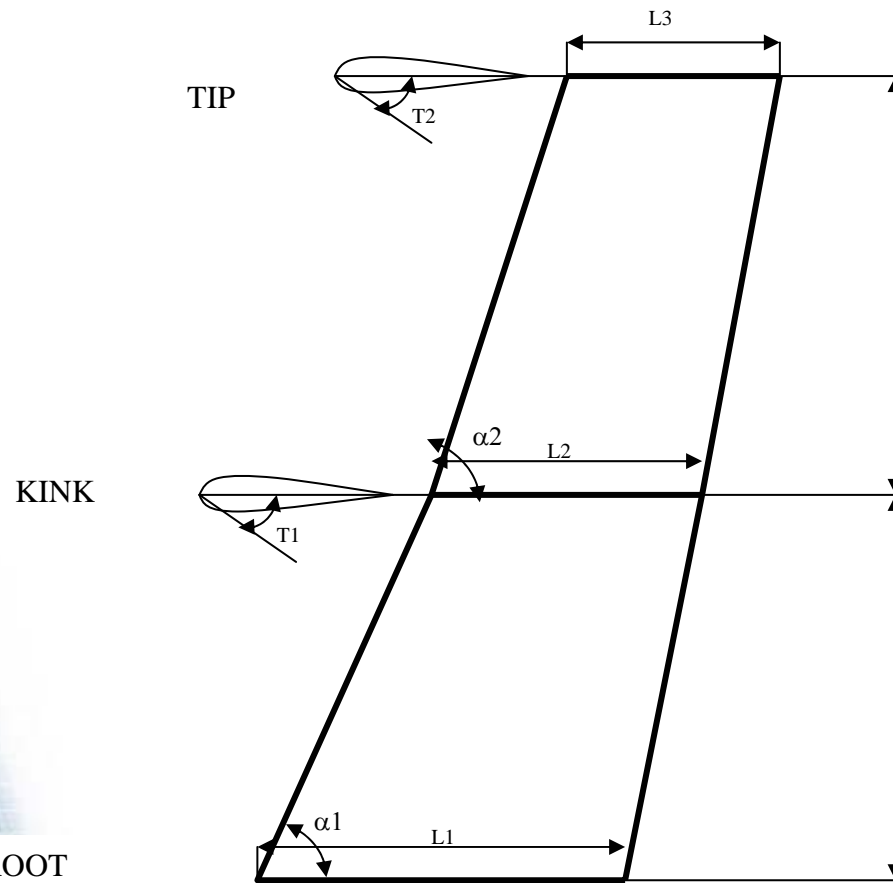




Parameterisation

- The 3D wing parametric model is defined inside CATIA V5:
 - 1) **9** geometric variables parameterise the planform
 - **2) 30** variables parameterise the shape of 3 main sections
 - **3) 26** variables parameterise internal shape and position of the flaps

Parameterisation



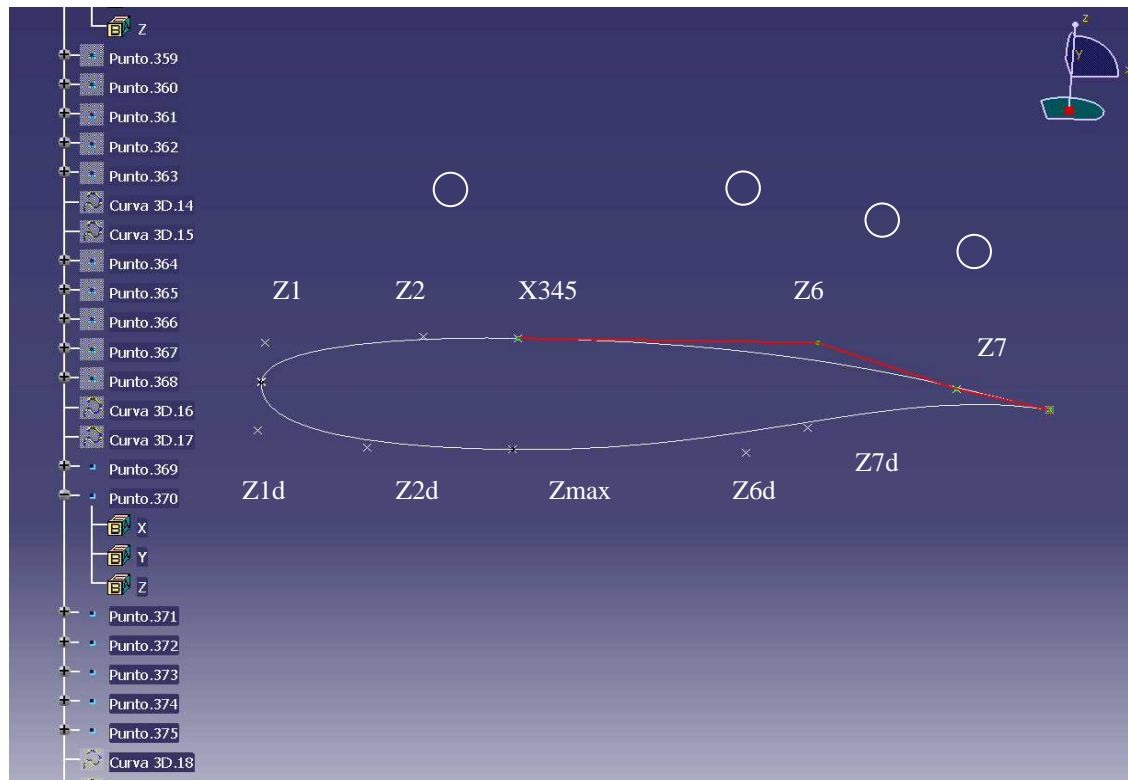
1) PLANFORM PARAMETERISATION

- L1:** length of root section (500mm)
- L2:** length of kink section (300mm)
- L3:** length of tip section (150mm)
- H1:** root-kink distance (500mm)
- H2:** kink-tip distance (1000mm)
- $\alpha 1$** root angle (55°)
- $\alpha 2$** kink angle (60°)
- t1** twist angle of kink section (0°)
- t2** twist angle of tip section (0°)

H1 These parameters were kept constant

ROOT

Parameterisation

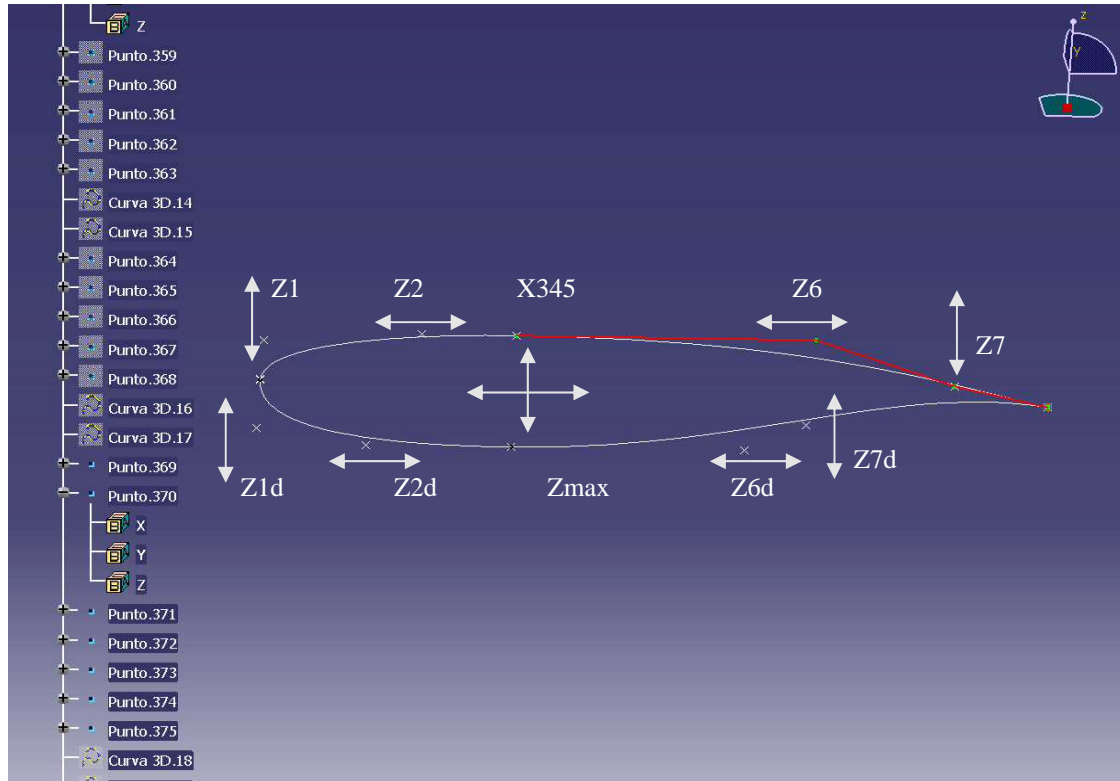


**2) SMOOTH SHAPE
PARAMETERISATION**

3 main sections (root,kink,tip)

4 NURBS curves (of 4 control points) for each section

Parameterisation



2) SMOOTH SHAPE PARAMETERISATION

3 main sections (root,kink,tip)

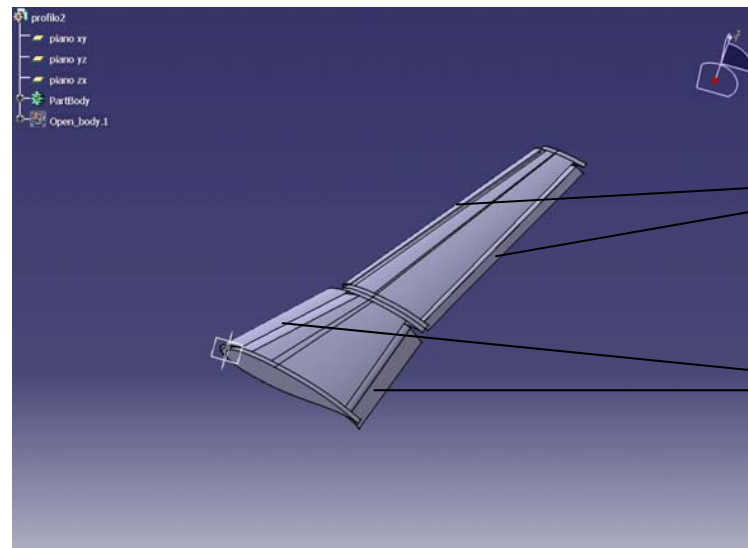
4 NURBS curves (of 4 control points) for each section

10 geometric variables for each section = **30 total sections shape variables**



Parameterisation

3) DEPLOYED POINT PARAMETERISATION



13 variables

13 variables

2 couples of slats and flaps (**2 sets of 13 variables**)

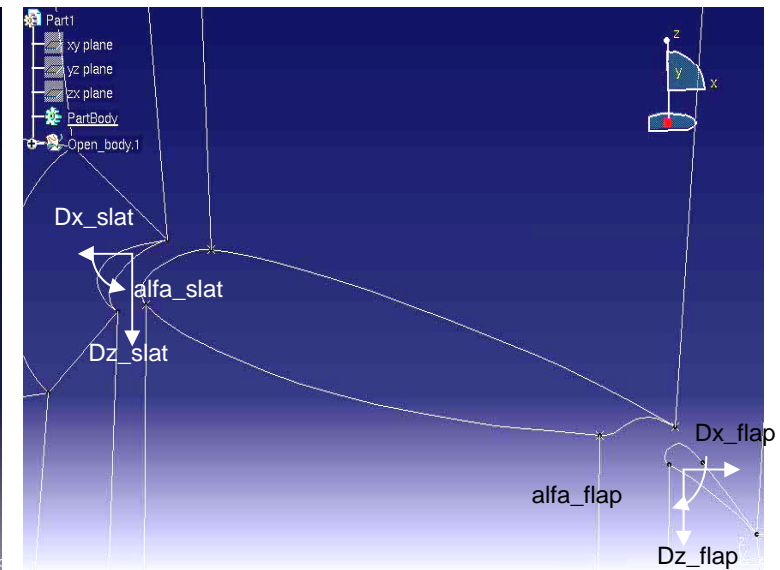
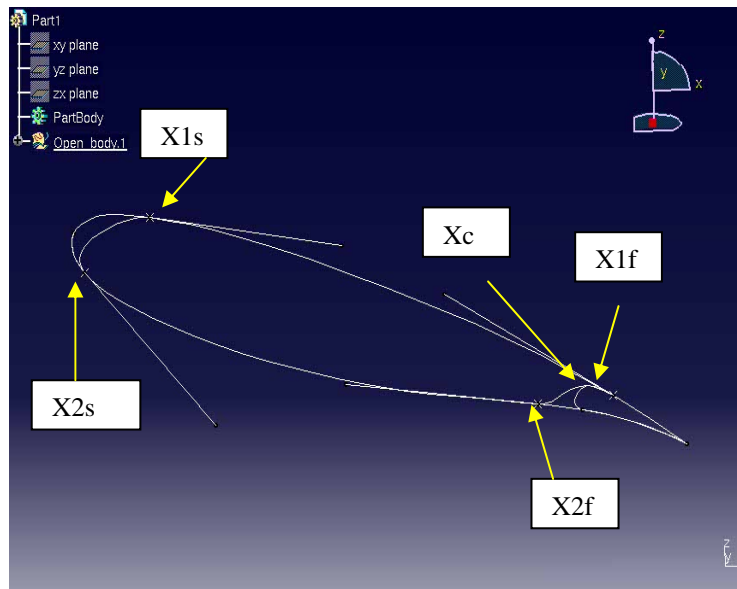
Each set corresponds to:

7 internal side shape variables

6 position variables

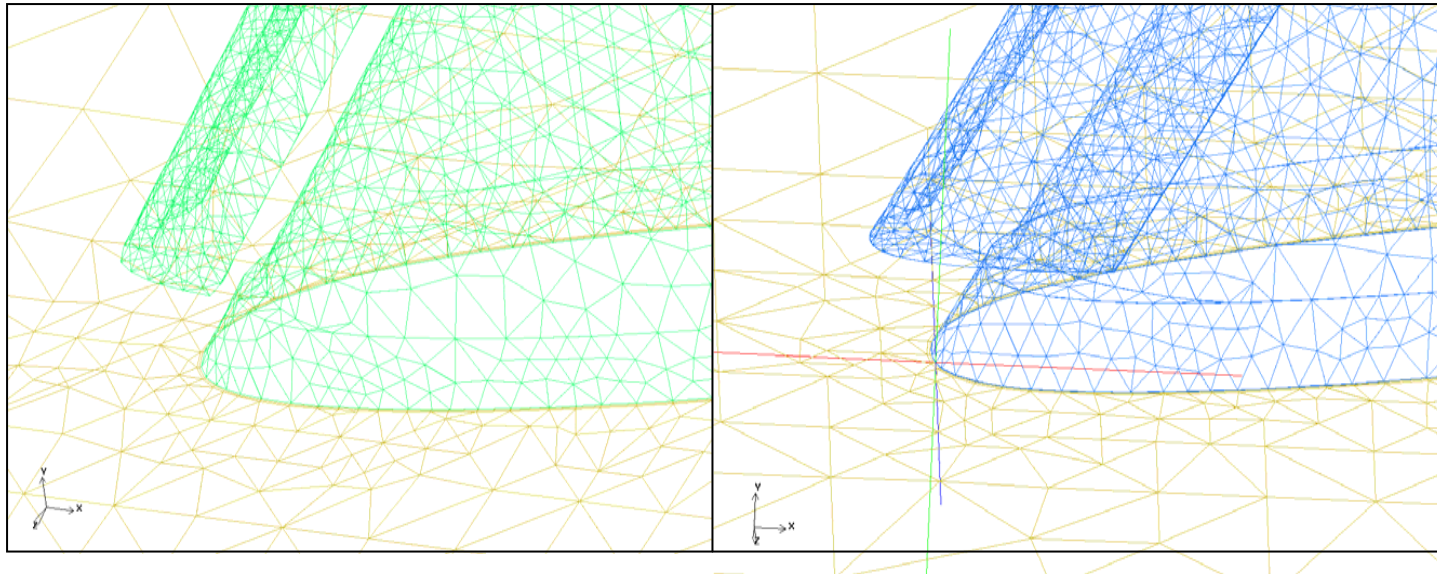
Parameterisation

3) DEPLOYED POINT PARAMETERISATION



internal sides of slat and flap are defined by **conic curves (7 var.)**
position of slat and flap is defined by **6 geometric variables**

Mesh generation



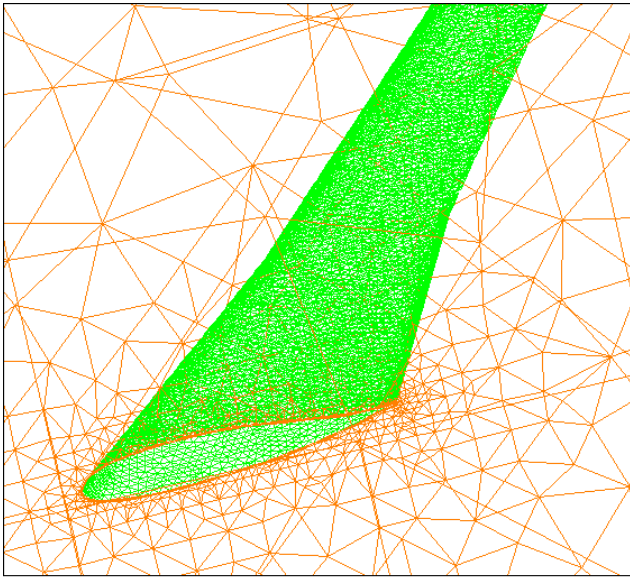
Catia → model file → **ICEM** → mesh file

Unstructured mesh: **tetrahedral** cells (about 500,000), 3 **prism** layers

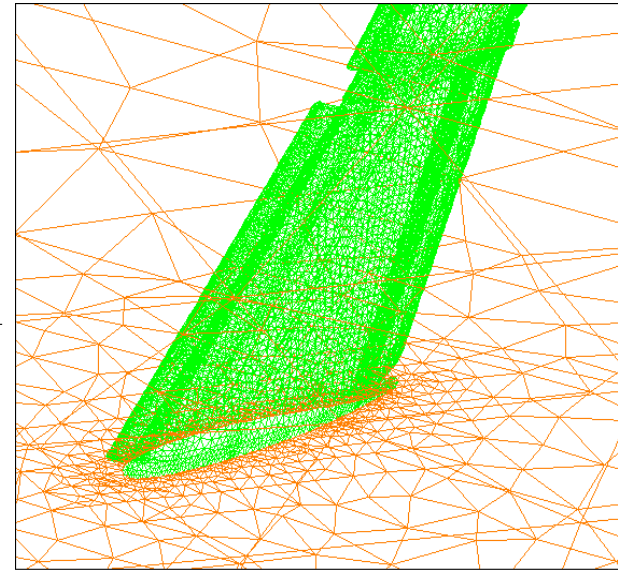
-Mesh conserves topology and accuracy at geometry variations

-Mesh is automatically updated for each configuration

Flow simulation



STARCD
PROSTAR 3.10
9-SEP-03
VIEW
0.015
1.000
0.607
ANGLE
14.705
DISTANCE
483.138
CENTER
453.026
1995.660
956.478
QHIDDEN PLOT



STARCD
PROSTAR 3.10
24-JUN-03
VIEW
-0.492
0.570
1.000
ANGLE
-6.413
DISTANCE
420.952
CENTER
-1027.11
1755.397
2626.219
QHIDDEN PLOT



ICEM → mesh file → STARCD → fitness values

- y^+ values are regular (30-150)
- each simulation (1 mission point for 1 configuration) requires about 2hs



Optimisation: Game Theory

- In a problem of minimisation of two functions $f_A(x,y)$ and $f_B(x,y)$ with $(x,y) \in A \times B$:
- $A \times B$ is defined as space of the rational strategies
- A is considered as **Player A**, he can play strategies x
- B is considered as **Player B**, he can play strategies y
- In a **Competitive Game**:
- Player A have to minimise function $f_A(x,y)$
- Player B have to minimise function $f_B(x,y)$



Optimisation: Game Theory

- The strategy $(x^*, y^*) \in A \times B$ is defined as a **Nash equilibrium** if and only if:

$$\begin{cases} f_A(x^*, y^*) = \inf_{x \in A} f_A(x, y^*) \\ f_B(x^*, y^*) = \inf_{y \in B} f_B(x^*, y) \end{cases}$$



Optimisation: Nash/Simplex algorithm

Simplex1 is run (Player 1)

Obj.: min. f_1

Var: \mathbf{X} , \mathbf{Y}_0 fixed

Simplex2 is run (Player 2)

Obj.: min f_2

Var: \mathbf{Y} , \mathbf{X}_0 fixed

Calculation of f_1 for every config.

n_1 iterations

Best $\mathbf{X} = \mathbf{X}_1$ is found

Calculation of f_2 for every config.

n_2 iterations

Best $\mathbf{Y} = \mathbf{Y}_1$ is found

Simplex1 is run (Player 1) again

Optimise \mathbf{X} with \mathbf{Y} fixed to \mathbf{Y}_1

Simplex2 is run (Player 2) again

Optimise \mathbf{Y} with \mathbf{X} fixed to \mathbf{X}_1

A converged optimized solution $(\mathbf{X}_N, \mathbf{Y}_N) = (\mathbf{X}_{N-1}, \mathbf{Y}_{N-1})$ is found



Optimisation: Nash/Simplex algorithm

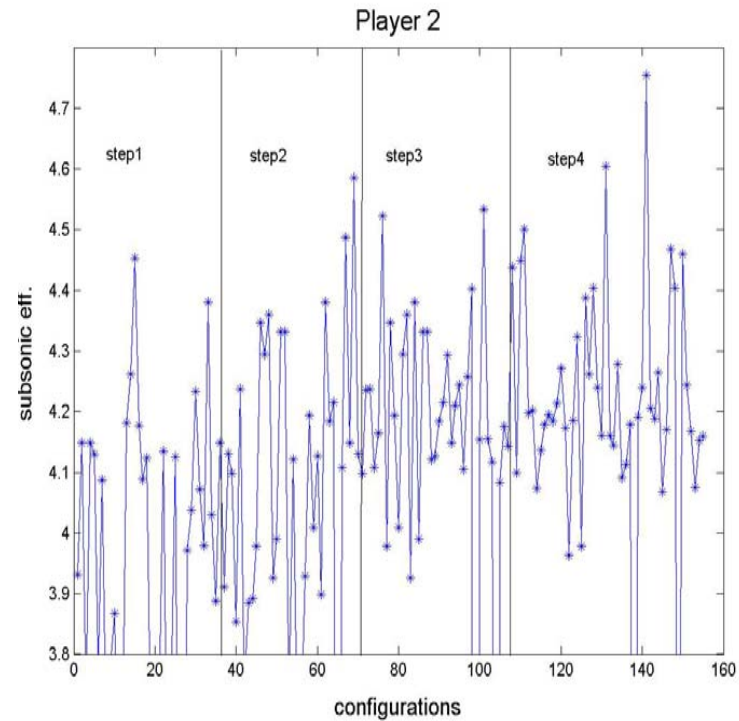
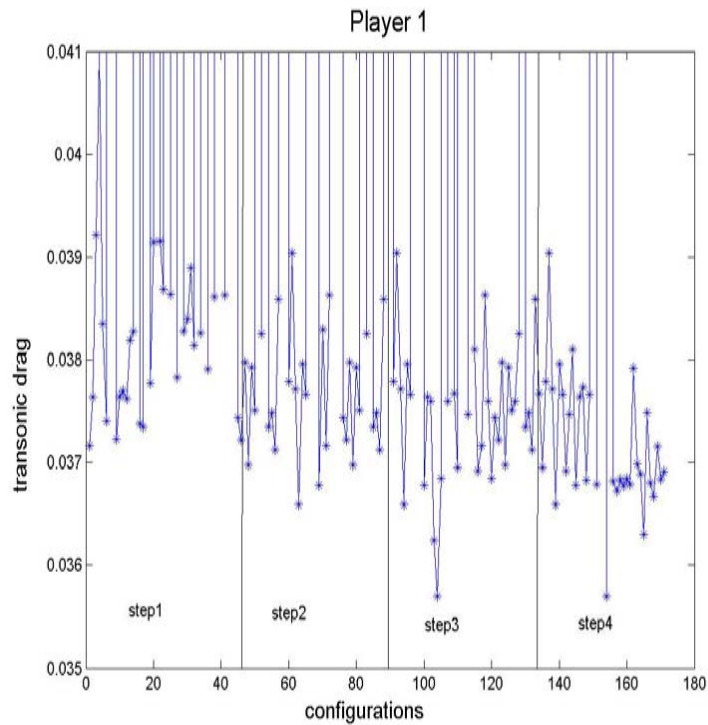
Player 1:

- the variables x are the 30 smooth sections shape ones
- the function $f1$ is the transonic point drag to minimise

Player 2:

- the variables y are the 26 internal shape and position ones
- the function $f2$ is the subsonic point efficiency to maximise

Optimisation results



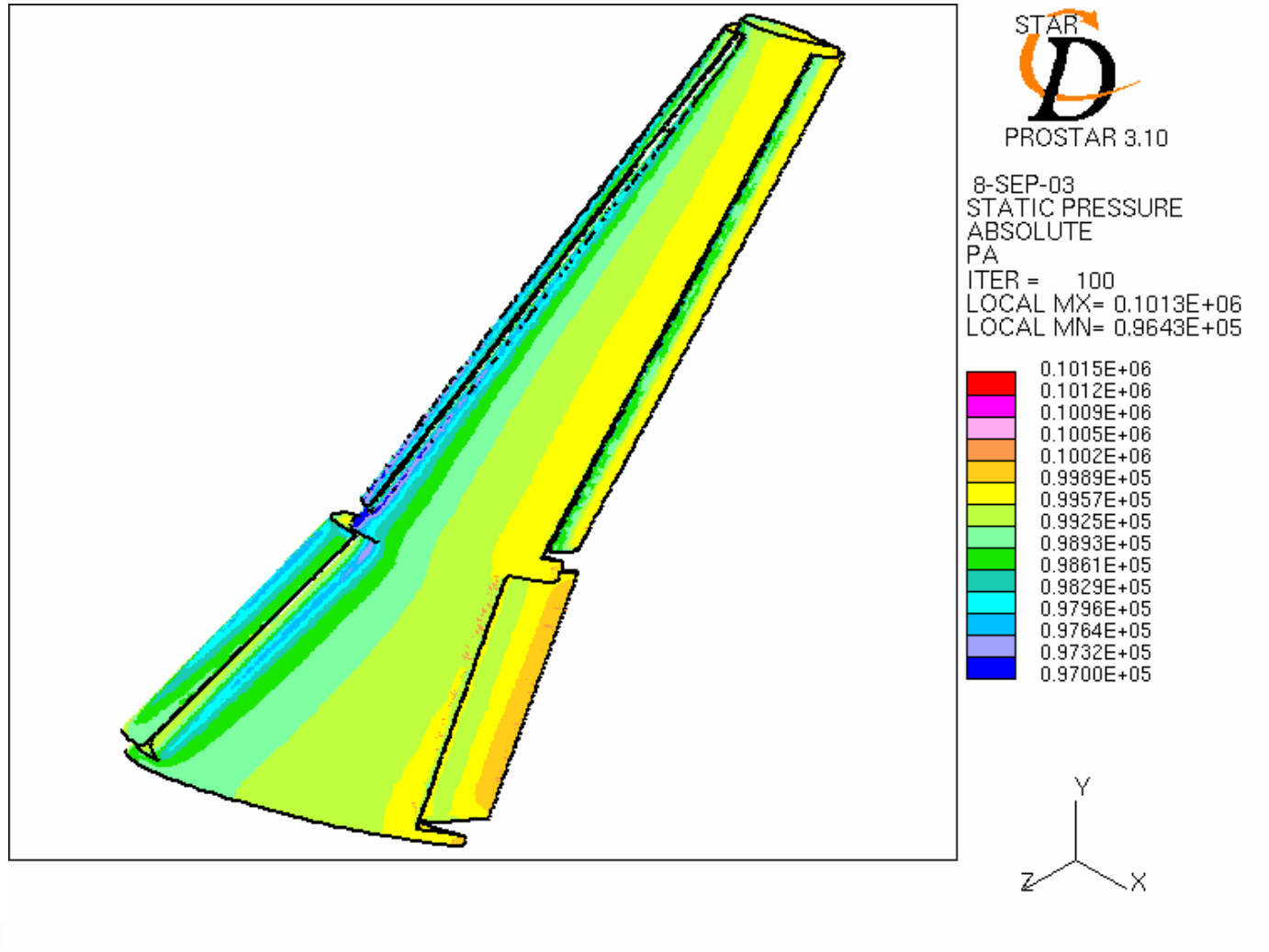
Player 1: 4 steps of 46 Simplex iterations

Player 2: 4 steps of 37 Simplex iterations

Total of about **350** simulations



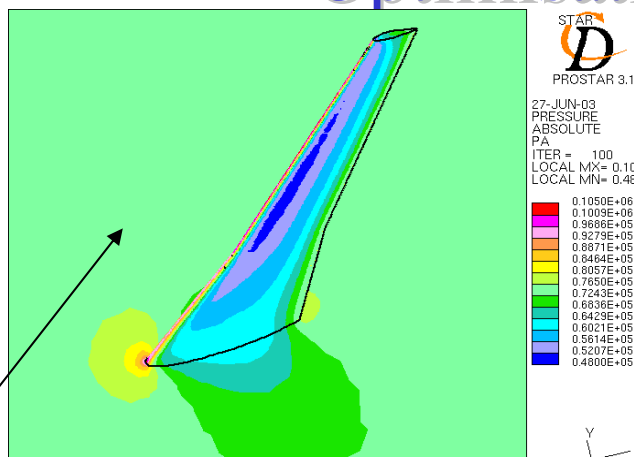
Optimisation results



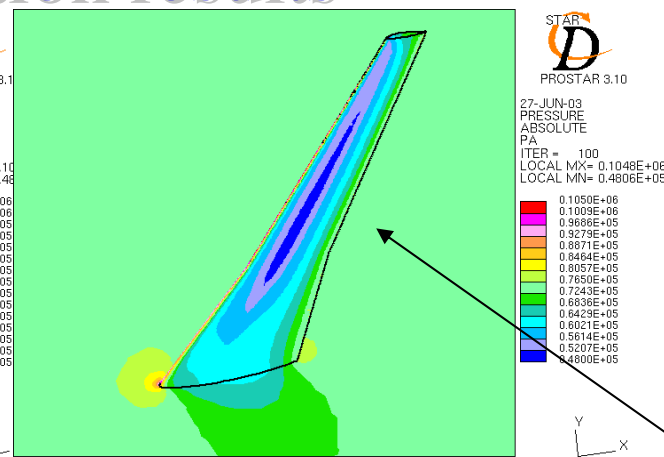


Optimisation results

Original
config.

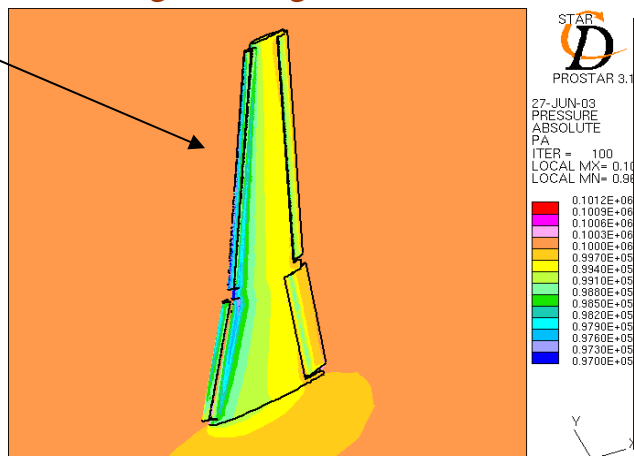


Original Drag = **0.0371**

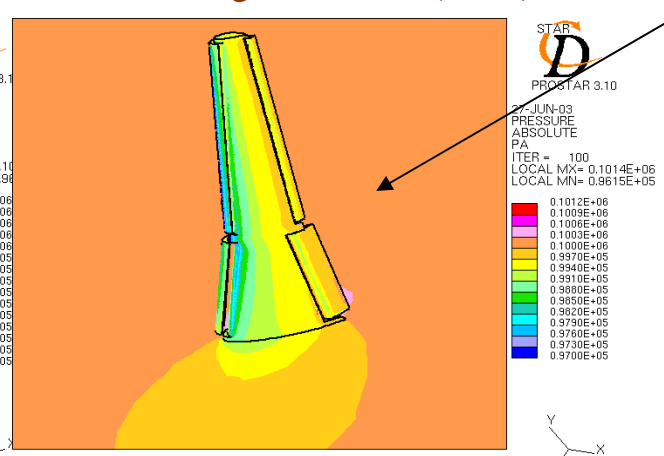


Best Drag = **0.0357 (-4%)**

Best
config.



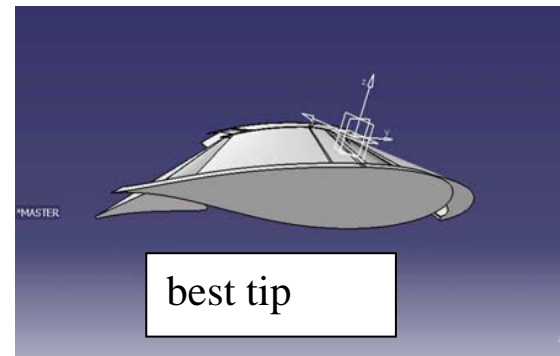
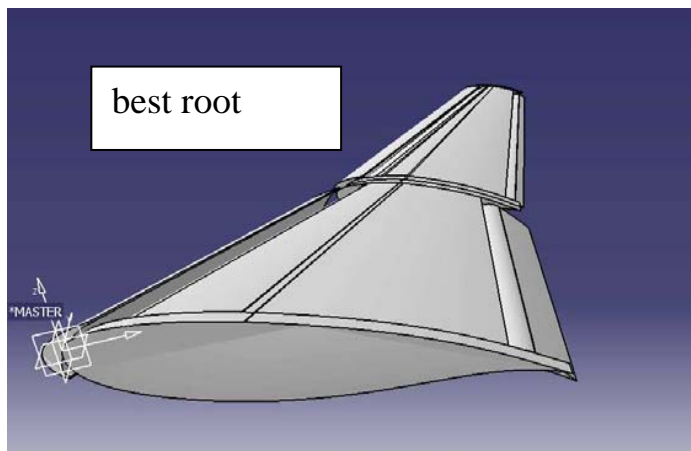
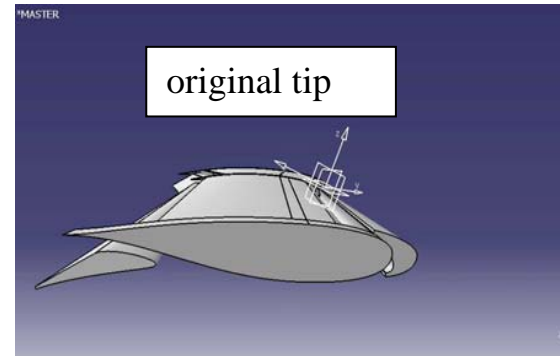
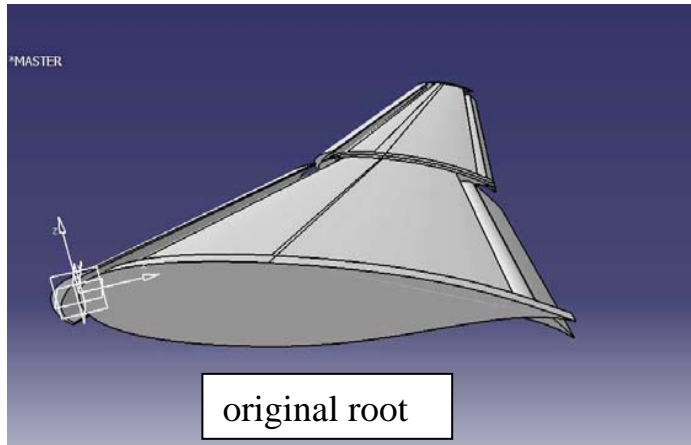
Original Efficiency = **3.93**



Best Efficiency = **4.75 (+30%)**



Optimisation results





Conclusions

ADVANTAGES OF CAD PARAMETERISATION

- any parametric CAD model can be integrated in the optimisation loop
- CAD tools allows the definition and parameterisation of complex design
(like the multi-point working wing)

EFFICIENCY OF GAME THEORY COMBINED WITH GA

- Nash/Simplex algorithm proved its efficiency if designer need a unique solution to be obtained by few simulations



Acknowledgements

- The first author thanks:
- DASSAULT AVIATION for the financial support to the research activity at University of Trieste
- ESTECO Ltd. for the use of modeFrontier